A DIVISION OF GENERAL DYNAMICS CORPORATION SAN DIEGO, CALIFORNIA

SPEC NO.: ZD. 22-324			DATE:		H	
CUSTOMER Federal A	viation Agency		ACL 14,017			
CHANGE NO: 13		Ь	MODEL: 22-2	21(#3)	(Conv	wair "880"
	ion Administrative Lagazine Rack)	Change	e (Revisio	n to Le	vator	ry Coat
ORIGIN: Convair in	itiated			44		
REASON FOR CHANGE:	To clarify the int	ent of	the speci	lficatio	on.	
EFFECT	ON WEIGHT *	Name and Associated	E F	FECT ON	BALAN	CE '
GUAR, WT. EMPTY	OPER, WT, EMPTY					
0	0			ν.	0	INCH LB.
EFFECT ON GUARANTEED	PERFORMANCE: * Non	e				
ACCEPTANCE OF THIS CH	LL BE ACCUMULATED AND FUTURE CHANGE PROPOSE HANGE IS DEPENDENT UPOTHE FOLLOWING CHANGES.	SAL Ni	Res			Jun 12/W
LATEST DATE OF ACCEPTA	ANCE:	AIRPLAN	NES AFFECTE	D:	Maintenance State of the Communication of the Commu	
SPECIAL PROVISIONS:		RECURRI	ON PRICE PE ING: URRING:			
ACCEPTED:		CONVAIR	, A DIVISION OF	FGENERAL	DYNAMI	CS CORP.
BY:					37	F 97.24
DATE:	5	4)	A.		305	1 VAIN, 2 DE 0 17 17 A

Title: Specification Administrative Change (Revision to Lavatory

Coat Hook, and Magazine Rack).

Origin: Convair initiated

Reason for Change: To clarify the intent of the specification.

Description of Change:

Page 89, Paragraph 3.19.2.2.3 MISCELLANEOUS LAVATORY EQUIPMENT:

Revise below item as follows:

From: "One coat hook (folding-type on lavatory door)"

To: "One coat hook"

Page 91, Paragraph 3.19.2.7.8 MAGAZINE RACKS:

Revise the sentence as follows:

From: "Magazine racks shall be provided in the forward and aft

entrance areas."

To: "Magazine racks shall be provided in the forward entrance

area."

A DIVISION OF GENERAL DYNAMICS CORPORATION SAN DIEGO, CALIFORNIA

SPEC NO.: ZD- 22-32	4	- 11	DATE:			
CUSTOMER: Pederal	Aviation Agency	_	MCL 16,014 DTD			
CHANGE NO: 12		VA.	MODEL: 22-21(#3) (Gonvair "880"			
TITLE: Specificat Improvemen	ion Administrative it Changes)	Chang	e (Miscellaneous Design			
ORIGIN: Convair in	itiated					
REASON FOR CHANGE:	Product improvemen	t	5 34			
	T ON WEIGHT *		EFFECT ON BALANCE *			
GUAR, WT. EMPTY	OPER. WT. EMPTY		O INCH LB.			
EFFECT ON GUARANTEE	D PERFORMANCE: * None	:0				
TOTALS REFLECTED IN ACCEPTANCE OF THIS	WILL BE ACCUMULATED AN I A FUTURE CHANGE PROP CHANGE IS DEPENDENT UP F THE FOLLOWING CHANGE	OSAL,	ENGINEERING APPROVAL			
LATEST DATE OF ACCEP	TANCE:	AIRPL	ANES AFFECTED:			
SPECIAL PROVISIONS:		RECUR	CT ON PRICE PER AIRPLANE: RRING: ECURRING:			
ACCEPTED:		CONVA	AIR, A DIVISION OF GENERAL DYNAMICS CORP.			
BY:						
DATE:	ISTO E		CONVAIR, S.D. 6-1217A			

Federal Aviation Agency Change No. 12

Page 1 of 2

Title: Specification Administrative Change (Miscellaneous Design

Improvement Changes)

Origin: Convair initiated

Reason for Change: Product improvement

Description of Change:

Page 3a, GENERAL ARRANGEMENT:

Replace above illustration in the specification with Enclosure (A) which shows Krueger Flap installation between inboard pylon and fuselage, and trailing edge inboard flap extension.

Page 28, Paragraph 3.10.1.3 RUDDER SYSTEM:

Insert the following sentence after the third sentence:

"In addition a hydraulically-operated servo unit powered from one hydraulic system shall be installed in the pedal linkage, in the nose wheel well, to reduce rudder pedal forces."

Page 29, Paragraph 3.10.2.1 LIFT AND DRAG INCREASING DEVICE SYSTEMS:

Delete the paragraph and substitute the following:

"LIFT AND DRAG INCREASING DEVICE SYSTEMS: The trailing edge flaps and leading edge slats and flaps shall be operated by hydraulic power controlled by a single lever on the pilot's pede~ stal. The leading edge slats and flaps shall be operated through torque tubes from a gear box which is powered by two hydraulic motors, one motor connected to each hydraulic system. The trailing edge flaps shall be operated through torque tubes from a gear box which is powered by either of two motors, one motor connected to each hydraulic system. By means of "no-back" type devices incorporated in the screw jacks, each control system shall be irreversible to prevent flap motion from airload. Full trailing edge flap deflection and leading edge slat and flap extension shall be obtainable from either hydraulic system. A monitoring system shall be provided to preclude asymmetrical positioning of the trailing edge flap in the event of mechanical failure. pair of lights, one amber and one green, shall be installed on the center instrument panel for each right and left hand set of leading edge devices (slats and flaps). Lights shall illuminate "green" when the devices have fully extended and shall illuminate "amber" when the devices are in transit."

Federal Aviation Agency Change No. 12

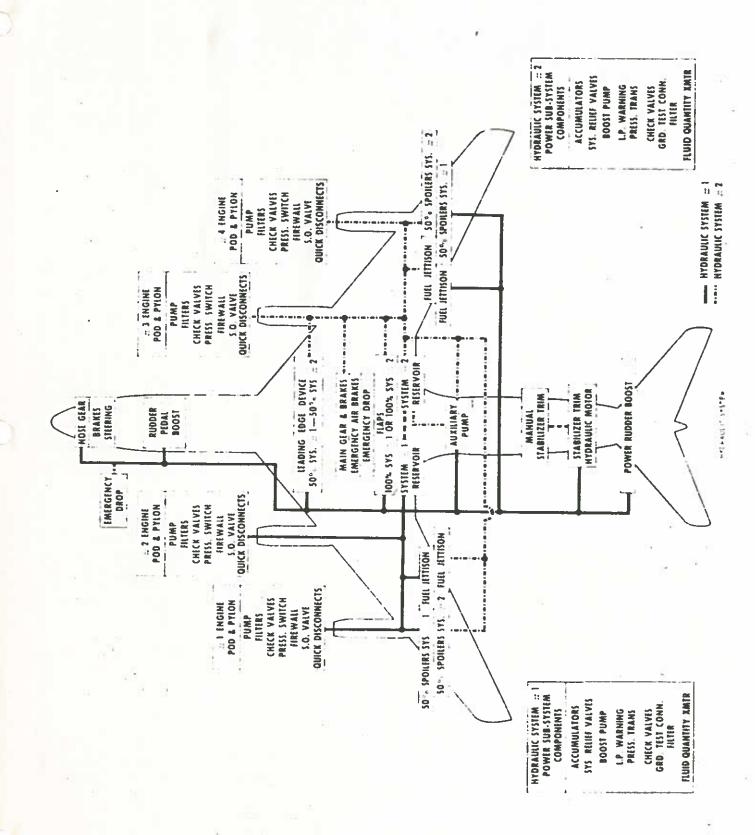
Page 2 of 2

Page 52a, HYDRAULIC SYSTEM:

Replace the above illustration in the specification with Enclosure (B) which shows 100 percent flap power available from either No. 1 or No. 2 Systems, and rudder pedal boost.

Enclosure: (A) One copy Page 3a - GENERAL ARRANGEMENT (Revised)

Enclosure: (B) One copy Page 52a - HYDRAULIC SYSTEM (Revised)



3000

CONVAIR

A DIVISION OF GENERAL DYNAMICS CORPORATION SAN DIEGO, CALIFORNIA

SPEC NO.: ZD- 22-324	DATE:
CUSTOMER: Federal Aviation Agency	MCL 16,016 DTD
CHANGE NO: 11	MODEL: 22-21 (#3) (Convair "88
TITLE: Specification Administrative 3.2.2.1)	ve Change (Revision to Paragraph
ORIGIN: Convair initiated	
REASON FOR CHANGE: Specification clari	lfication.
EFFECT ON WEIGHT *	EFFECT ON BALANCE *
GUAR, WT. EMPTY OPER, WT. EMPTY	
0 0	O INCH LB.
EFFECT ON GUARANTEED PERFORMANCE: *	
None	
* NEGLIGIBLE CHANGES WILL BE ACCUMULATED AND TOTALS REFLECTED IN A FUTURE CHANGE PROPOSE ACCEPTANCE OF THIS CHANGE IS DEPENDENT UPO PRIOR ACCEPTANCE OF THE FOLLOWING CHANGES:	SAL CONFORMING (47)
LATEST DATE OF ACCEPTANCE:	AIRPLANES AFFECTED:
SPECIAL PROVISIONS:	EFFECT ON PRICE PER AIRPLANE: RECURRING: NON-RECURRING: TOTAL:
ACCEPTED:	CONVAIR, A DIVISION OF GENERAL DYNAMICS CORP.
BY:	*
DATE:	#

Federal Aviation Agency Change No. 11

Title: Specification Administrative Change (Revision to Paragraph

3.2.2.1)

Origin: Convair initiated

Reason for Change: Specification clarification.

Description of Change:

Page 9, Paragraph 3.2.2.1 COLOR CODING:

Change the paragraph as follows:

"COLOR CODING: All sections of plumbing shall be color coded and banded in accordance with Military Standards and, in addition, marking shall indicate the direction of flow and the purpose of the line. The part number of each tubing assembly shall be identified. Color bands and part number identifications shall be plastic tape in accordance with Standard AND 10375."

To: "COLOR CODING: All sections of plumbing, except engine and within fuel tanks, shall be color coded and banded. All plumbing shall indicate the direction of flow and the purpose of the line. The part number of each tubing assembly shall be identified. Color bands shall be plastic tape in accordance with Standard AND 10375."

DIVISION OF BEHERAL DYNAMICS COMPURATION. SAN DIE GO. CALIFORNIA

SPEC NO. ZD- 22-324			DATE			
CUSTOMER. Federal Av	iation Agency		MCL 16,015 D	TD		
CHANGE NO: 10			MODEL 22-21(#3)	(Convair "880"		
TITLE Specificat Lights)	ion Administrative	e Char	nge (Landing Gea	r Pin Indicator		
ORIGIN Convair in	itiated					
REASON FOR CHANGE T	o specify that sul	ject	indicator light	s are installed.		
EFFECT	ON WEIGHT .		EFFECT	N BALANCE		
GUAR, WT. EMPTY	OPER. WT. EMPTY					
0	0		0 INCH LB.			
EFFECT ON GUARANTEED P	ERFORMANCE: *	9				
* NEGLIGIBLE CHANGES WIL TOTALS REFLECTED IN A ACCEPTANCE OF THIS CHAPRIOR ACCEPTANCE OF T	FUTURE CHANGE PROPOS ANGE IS DEPENDENT UPO	SAL	Aleady	19 Mil Maligh		
			ENGI	NEERING APPROVAL		
LATEST DATE OF ACCEPTA	NCE.	AIRPL	ANES AFFECTED:			
SPECIAL PROVISIONS		RECUR NON-RI TOTAL	EWRRING:			
ACCEPTED:			IR, A DIVISION OF GENER			
5Y						
DATE:				(0) 994 1, 1 @ (04217)		

Federal Aviation Agency Change No. 10

Title: Specification Administrative Change (Landing Gear Pin

Indicator Lights)

Origin: Convair initiated

Reason for Change: To specify that subject indicator lights are

installed.

Description of Change:

Page 23, Paragraph 3.8.1.5 LANDING GEAR SAFETY LOCK PINS:

Add the following to the end of the paragraph:

"Indicator warning lights shall be provided in the flight engineer's panel and shall be illuminated when the "safety lock" pins are inserted in the gear."

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SPEC NO.: ZD- 22-324			DATE.		
CUSTOMER Federal A	viation Agency		MCL 16,057	DTD 2	March 1961
CHANGE NO: 9			MODEL: 22-2	1 (#3) (Co	nvair "880"
TITLE Main Tires,	Fabric Reinforced,	Inst	allation of		
ORIGIN Convair init		;			
EFFECTO	DN WEIGHT .		EFI	ECT ON BALA	NCE *
	OPER, WT. EMPTY				
+23.0 1b	+23.0 lb			+20,194	INCH LB.
EFFECT ON GUARANTEED P	ERFORMANCE: None				
* NEGLIGIBLE CHANGES WILL TOTALS REFLECTED IN A ACCEPTANCE OF THIS CH. PRIOR ACCEPTANCE OF T	FUTURE CHANGE PROPOS ANGE IS DEPENDENT UPO	SAL N	-Jul	Curly &	Hale, Undit
LATEST DATE OF ACCEPTA	NCE.	AIRPL	ANES AFFECTED	D;	
SPECIAL PROVISIONS.		RECUR NON-RI	T ON PRICE PER RING: CURRING:		
ACCEPTED:		CONVA	IR, A DIVISION OF	GENERAL DYNA	AMICS CORP.
ВУ					
DATE:	- 10-16-7				file Calling to 6-1, 17 a

Federal Aviation Agency Change No. 9

CONVAIR: SD

Title: Main Tires, Fabric Reinforced, Installation

Origin: Convair initiated

Reason for Change: Product improvement

Description of Change:

Page A-13, APPENDIX I-C, LANDING GEAR EQUIPMENT:

Revise the below item under "Description" as follows:

From: "8 Tire, Main Wheel (39 x 13)
Type VII 22 Ply Rating,
Skid Depth .25-inches)

U. S. Rubber"

To: "8 Tire, Main Wheel (39 x 13) (Type VII, 22-Ply Rating, Fabric-Reinforced Tread, Skid Depth .35-inch)

U. S. Rubber"

Effect on Weight Empty: +23.0 pounds

Effect on Balance: +20,194 inch-pounds

Effect on Performance: None

A HIRLAGON OF SENTENAL DENAMING CORPORATION

MCL 16,055 DTD 20 February 19 MODEL 22-21(#3) (Convair "880"
MODEL 22-21(#3) (Convair "880"
ton Office
EFFECT ON BALANCE •
EFFECT ON BALANCE
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LANES AFFECTED:
CT ON PRICE PER AIRPLANE RRING: RECURRING.
AIR, A DIVISION OF GENERAL DYNAMICS CORP.

Federal Aviation Agency Change No. 8

Page 1 of 2

Title: Seating Capacity, Increase of

Origin: FAA request to Convair Washington Office

Reason for Change: As above

Description of Change:

Page 1, Paragraph 1.1 BASIC TYPE:

In the sixth line, change "12" to "40".

Page 3b, INTERIOR ARRANGEMENT:

Replace the above illustration with Enclosure (A) which shows the effects of this proposal.

Page 6, Paragraph 3.1.2.2 TYPICAL LOADING SUMMARY:

Revise the below item under "Fixed Useful Load" as follows:

From: "Life Vests (18) 32 lb"

To: "Life Vests (46) 82 lb"

Revise the below items under "Space Limit Payload" as follows:

From:	"Passenger (12) at 165Lb Baggage @ 40 lb/passenger Cargo	Total	1,980 lb 480 lb 8,150 lb 10,610 lb
To:	"Passengers (40) at 165-1b		6,600 lb

To: "Passengers (40) at 165-1b 6,600 1b Baggage @ 40 lb/passenger 1,600 lb 7,030 lb Total 15,230 lb"

Page 85, Paragraph 3.19.1 GENERAL:

In the second line, change "12" to "40".

Page 96, Paragraph 3.19.7.1 LIFE RAFTS:

Revise subparagraphs "b." and "c." as follows:

Federal Aviation Agency Change No. 8

Page 2 of 2

"b. One on each side of the aisle on the floor in passen-From: ger compartment aft of the emergency escape hatches.

One on the floor, aft of the left hand passenger seat.

"b. To: Three on the floor, aft of last passenger seat row, one on right side of the aisle and two on left side of the aisle."

Page A-2, APPENDIX I-B, CUSTOMER FURNISHED - CUSTOMER INSTALLED:

Revise the below item under "description" as follows:

From: " 18 Life Vests 32.0"

"*46 Life Vests To: 82.0"

Page A-14, APPENDIX I-C, FURNISHINGS:

Revise the below items under "Description" as follows:

' 6 Seats, Double (First Class) 8 Tray, Integral Folding Food 12 Belt, Passenger Safety" From:

"10 10 Seats, Passenger, Double L.H. (First Class) 10 Seats, Passenger, Double R.H. (First Class) To:

36 Tray, Integral Folding Food 40 Belt, Passenger, Safety"

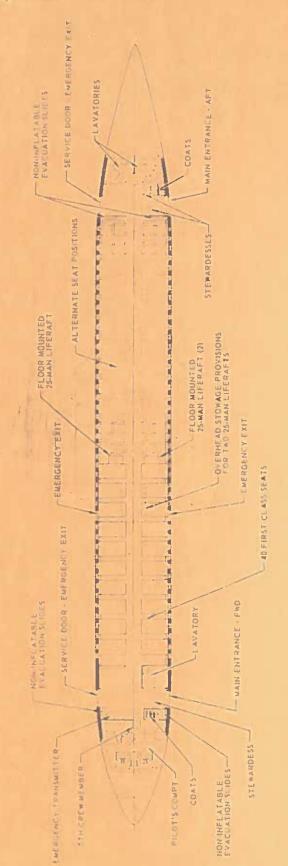
Enclosure: (A) One copy Page 3b - INTERIOR ARRANGEMENT (Revised)

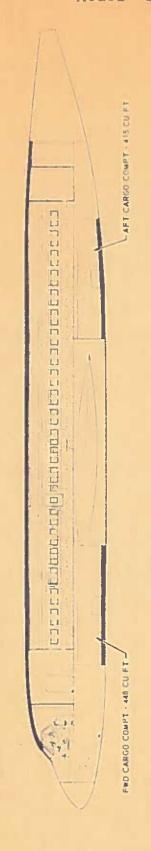
Effect on Weight

Effect on Balance

+890.0 pounds +594,568 inch-pounds Weight Empty: Operating Weight Empty: +940.0 pounds +695,554 inch-pounds

Effect on Performance: None





6027A

INTERIOR ARRANGEMENT
40 FIRST CLASS PASSENGERS

FAA

A DIVISION OF GENERAL DYNAMICS CORPORATION SAN DIFGO, CALIFORNIA

SPEC NO.: ZD- 22-324			DATE:				
CUSTOMER: Federal A	viation Agency	MCL 16,013 DTD					
CHANGE NO 6			MODEL: 22-21(#3)	(Conva	lr "880"		
TITLE: Specificat: Seat Back I		Chan	ge (Revision to Pa	assenge	[3		
ORIGIN: Convair in:	itiated				estado e esta		
	To make specificat back positioning c		anguage agree with lity.	n actual	l seat		
EFFECT	ON WEIGHT *		EFFECT ON	BALANCE	•		
GUAR. WT. EMPTY	OPER. WT. EMPTY						
0	0			0	INCH LB.		
EFFECT ON GUARANTEED F	PERFORMANCE: * None						
* NEGLIGIBLE CHANGES WILL TOTALS REFLECTED IN A ACCEPTANCE OF THIS CH. PRIOR ACCEPTANCE OF T	FUTURE CHANGE PROPOS ANGE IS DEPENDENT UPO	SAL N	Attenday	FRING APP			
LATEST DATE OF ACCEPTA	NCE:	AIRPL	ANES AFFECTED:				
SPECIAL PROVISIONS.		RECUR NON-RI	TON PRICE PER AIRPLAIREING:ECURRING:				
ACCEPTED:	Are a some to a consequence of the consequence of t	CONVA	AIR, A DIVISION OF GENERAL	DYNAMICS	CORP.		
BY:							
DATE:				SONVA	4.5.U. 6.1217A		

Federal Aviation Agency Change No. 6

Title: Specification Administrative Change (Revision to Passenger

Seat Back Movement)

Origin: Convair initiated

Reason for Change: To make specification language agree with actual

seat back positioning capability.

Description of Change:

Page 86, Paragraph 3.19.1.1.6.1 SEAT BACK MOVEMENT:

Revise the paragraph to read as follows:

"Each reclining seat shall be designed so that pressure applied on the aft side of the seat back will override the recline control lock without additional manual operation, and fold the seat back forward to its normal upright position. This pressure on the seat back shall not exceed 25 pounds. The seat back shall fold to approximately 30 degrees forward of the vertical position by applying a load of not less than 30 pounds, or more than 35 pounds, at the top of aft side of the seat back without removing the center arm rest. The seat back may be further folded to approximately a horizontal position by removing the center arm rest. Seat back positions shall be as follows:

Normal Upright: 12 degrees aft of vertical

Recline: 38 degrees aft of vertical"



A DIVISION OF SENERAL DYNAMICS CORPORATION BAN DIEGO, CALIFORNIA

SPEC NO. 70-22-324		DATE:				
CUSTOMER Federal Av	CUSTOMER Federal Aviation Agency		MCL 16,012 DTD			
CHANGE NO 5			MODEL 22-21(#3)(Convair "880")			
TITLE Specifica Useful Lo	tion Administrativad Items)	ve Cha	nge (Revision	to Append	ix I-C	
ORIGIN Convair i	nitiated		V.			
REASON FOR CHANGE T	o clarify the inte	ent of	the specific	ation.		
EFFECT	ON WEIGHT *		EFFE	CT ON BALANCE	E*	
GUAR, WT. EMPTY	OPER, WT, EMPTY					
0	0			0	INCH LB.	
EFFECT ON GUARANTEED P	PERFORMANCE: * None				•	
* NEGLIGIBLE CHANGES WILL TOTALS REFLECTED IN A ACCEPTANCE OF THIS CHAPRIOR ACCEPTANCE OF T	FUTURE CHANGE PROPOS ANGE IS DEPENDENT UPO	SAL N	Juff	erogy/46 NGINEERING AI	PPROVAL	
LATEST DATE OF ACCEPTA	NCE:	AIRPL	ANES AFFECTED:			
SPECIAL PROVISIONS:		RECUR NON-R!	T ON PRICE PER A RING: ECURRING: :			
ACCEPTED:		CONVA	IR, A DIVISION OF GE	NERAL DYNAMIC	S CORP.	
8Y						
DATE				CONV	'AIR,5.0, 6-1217A	

Title: Specification Administrative Change (Revision to Appendix I-C

Useful Load Items)

Origin: Convair initiated

Reason for Change: To clarify the intent of the specification.

Description of Change:

Page A-15, APPENDIX I-C, OXYGEN EQUIPMENT:

Revise the two top items under "Description" as follows:

From: "3 *Cylinders, High Pressure
Oxygen (107 cu ft) (with
gages)
4 *Bottle, Portable Oxygen
(310-liter)
60.0

To: "3 *Cylinders, High Pressure Zep Aero ZC268-111-10 140.0 Oxygen (107 cu ft, with gages)

1 *Bottle, Portable Oxygen and Smoke Mask (310- Scott 6080-3 liter) 3 *Bottle, Portable Oxygen Scott 5500ClA-BF20B) (310-liter)

Add the following to the bottom of Page A-15:

"*Useful Load Items"

Page A-16, APPENDIX I-C, FIRE EXTINGUISHING EQUIPMENT:

Add the following to the bottom of Page A-16:

"*Useful Load Items"



A DIVISION OF GENERAL DYNAMICS CORPORATION SAN DIE GO, CALIFORNIA

SPEC NO. ZD- 22-324			DATE:		
CUSTOMER: Federal Avi	ation Agency		MCL 16,054 DTD		
CHANGE NO: 4	4		MODEL: 22-21(#3) (Convair "88		
TITLE Main System	Hydraulic Pumps,	Chan	ge to		
ORIGIN: Convair ini	tiated				
REASON FOR CHANGE: P	Product improvemen	ı ft			
	,	10			
EFFEGT O	N WEIGHT .		EFFECT	ON BALANC	E *
GUAR, WT., EMPTY	OPER. WT. EMPTY				
+31.0 lb	+31.0 lb		+23	,091	INCH LB.
EFFECT ON GUARANTEED PE	ERFORMANCE: Non	ne			
NEGLIGIBLE CHANGES WILL TOTALS REFLECTED IN A F ACCEPTANCE OF THIS CHA PRIOR ACCEPTANCE OF TH	FUTURE CHANGE PROPOS NGE IS DEPENDENT UPOR	SAL .	Allen	they/	of chy
			EN	GINE EPING AF	PPROVAL
LATEST DATE OF ACCEPTAN	KCE:	AIRPLA	NES AFFECTED:		
SPECIAL PROVISIONS:			T ON PRICE PER AIR		
		NON-RE	RING:		
ACCEPTED: .		CONVA	R, A DIVISION OF GEN	ERAL DYNAMIC	s CORP.
BY:			,*		
DATE:				conv	a: (, p). 0-1217 a

Federal Aviation Agency Change No. 4

Title: Main System Hydraulic Pumps, Change to

Origin: Convair initiated

Reason for Change: Product improvement

Description of Change:

Page A-11, APPENDIX I-C, HYDRAULIC AND PNEUMATIC EQUIPMENT:

Revise the below item under "Description" as follows:

From: "4 Pump, Main System (Variable Displacement

Type) Ham. Standard"

To: "4 Pump, Main System (Variable Displacement

Type) Vickers AS-61695-L-2"

Effect on Weight Empty: +31.0 pounds

Effect on Balance: +23,091 inch-pounds

Effect on Performance: None

The following shall not appear in the Specification language:

The hydraulic pump will incorporate one electrical depressurization valve and one blocking valve. The bolt-on ports will be modified to suit the Model 22 installation, and the yoke blocked back to a 6 gpm rating in lieu of 10 gpm.

This proposal also deletes the four accumulators in the hydraulic return lines, since these accumulators are not required with the Vickers pump.



A DIVISION OF GENERAL DYNAMICS COMPORATION SAN DIEGO, CALIFORNIA

SPEC NO.: ZD- 22-324	DATE:
CUSTOMER: Federal Aviation Agency	MCL 16,011 DTD
CHANGE NO: 3	MODEL: 22-21(#3) (Convair "880
TITLE: Specification Administrat 3.10.1.5.1 and 3.11.8)	ive Change (Revision to Paragraphs
ORIGIN: Convair initiated	
REASON FOR CHANGE: To clarify the int	ent of subject paragraphs.
FFFECT ON WEIGHT *	EFFECT ON BALANCE *
GUAR, WT. EMPTY OPER, WT, EMPTY	•
0 0	O INCH LB.
Non NEGLIGIBLE CHANGES WILL BE ACCUMULATED AN TOTALS REFLECTED IN A FUTURE CHANGE PROPE	D
ACCEPTANCE OF THIS CHANGE IS DEPENDENT UPOPPIOR ACCEPTANCE OF THE FOLLOWING CHANGES	
LATEST DATE OF ACCEPTANCE:	AIRPLANES AFFECTED:
SPECIAL PROVISIONS:	RECURRING: NON-RECURRING: TOTAL:
ACCEPTED: BY:	CONVAIR, A DIVISION OF GENERAL DYNAMICS CORP.
DATE:	CONVAMINATION WITSON

Federal Aviation Agency Change No. 3

Title: Specification Administrative Change (Revision to Paragraphs

3.10.1.5.1 and 3.11.8)

Origin: Convair initiated

Reason for Change: To clarify the intent of subject paragraphs:

Description of Change:

Page 28, Paragraph 3.10.1.5.1 STABILIZER TRIM CONTROL:

Delete the last two sentences and substitute the following:

"A switch, guarded in the on position, shall be installed on the pedestal to shut off all hydraulic power to the stabilizer screw jack. A three-position momentary switch shall be installed on the pedestal which, after the hydraulic shutoff switch is moved to "OFF", operates the standby electrical-mechanical stabilizer trim control. In addition, hand cranks shall be installed on the pedestal for emergency manual stabilizer trim control."

Page 31, Paragraph 3.11.8 INSPECTION AND MAINTENANCE:

Delete the last sentence and substitute the following:

"Each engine QEC (quick engine change build-up) shall be interchangeable and replaceable as a complete assembly, except for reverser cascades, and certain accessories which may not be required on individual installations."

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SREC NO. ZD 22-324_			DATE:				
CUSTOMER Federal Av	CUSTOMER Federal Aviation Agency			DTD			
CHANGE NO 2			MODEL : 22-21(#3) (Conv	air "880")		
Specificat 3.17.1.7.1		e Cha	nge (Revisio	n to Para	graph		
ORIGIN Convair in	nitiated						
REASON FOR CHANGE:	To clarify the in	tent	of Paragraph	3.17.1.7	.1.		
BFFECT O	ON WEIGHT "		EFF	ECT ON BALA	NCE *		
GUAR, WT. EMPTY	OPER, WT. EMPTY						
0	0		,	0	INCH LB.		
EFFECT ON GUARANTEED P	ERFORMANCE: "						
	None						
NEGLIGIBLE CHANGES WILL TOTALS REFLECTED IN A ACCEPTANCE OF THIS CH PRIOR ACCEPTANCE OF T	FUTURE CHANGE PROPOS ANGE IS DEPENDENT UPOR	SAL N	gaten	Typ 1/31/6	GAPPROVAL		
LATEST DATE OF ACCEPTA	NCE:	AIRPL	ANES AFFECTED				
RECINON		RECURRING: NON-RECURRING: TOTAL:					
ACCEPTED:		CONV	AIR, A DIVISION OF	GENERAL DYN	AMICS CORP.		
BY							
DATE					The second		

Title: Specification Administrative Change (Revision to Paragraph

3.17.1.7.1)

Origin: Convair initiated

Reason for Change: To clarify the intent of Paragraph 3.17.1.7.1.

Description of Change:

Page 77, Paragraph 3.17.1.7.1 POWER CONTROLS:

Delete the paragraph and substitute the following:

"3.17.1.7.1 POWER CONTROLS: One "NORMAL" and one "ESSENTIAL" switch, located accessible to both pilots, shall control the ac and dc power to the radio equipment. The "NORMAL" switch shall control the weather radar, the marker beacon, selective call systems No. 1 and No. 2, if installed, glide slope No. 2, navigation instrument transformer No. 2, VHF navigation No. 2, radio compass No. 2, VHF communication receiver No. 2, VHF communication transmitter No. 2, HF communication No. 2, if installed, DMET No. 1 and No. 2, if installed, and ATC transponder No. 1 and No. 2, if installed. The "ESSENTIAL" switch shall control the VHF communication receiver No. 1, VHF communication transmitter No. 1, VHF navigation No. 1, glide slope No. 1, Radio compass No. 1, navigation instrument transformer No. 1, HF communication No. 1, if installed. The emergency dc bus, through suitable circuit breakers, shall power the audio selector panels, the public address amplifier, the interphone, the pilot compartment speaker, the flight data recorder, and the tape reproducer, if installed. No. 1 normal radio equipment shall be supplied by a separate ac and dc line. All No. 2 normal radio equipment shall be supplied by a second separate ac and dc line. Protection shall be provided for each line. The power supply, feeder wires, circuit protection for equipment which back up each other functionally shall be separated in a manner that will preclude operation failure of both if any one circuit protective device feeder wire or power supply fails. Illumination shall be provided for the circuit breaker

Federal Aviation Agency Change No. 1

Title: Specification Administrative Change (Revision to Auxiliary Landing Gear Tail Skid)

Origin: Convair initiated

Reason for Change: Product improvement

Description of Change:

Page 24, Paragraph 3.8.3 AUXILIARY LANDING GEAR (TAIL SKID):

Change the first line in the paragraph to read as follows:

From: "A retractable skid of"

To: "A skid or"